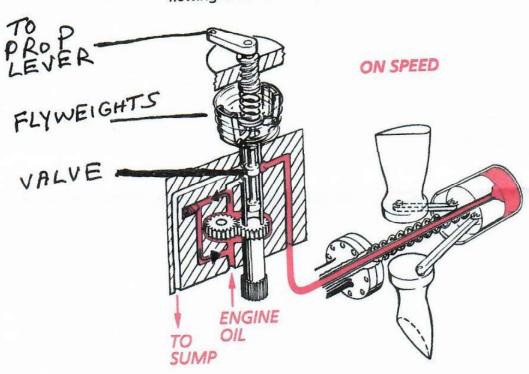
So how does all this result in constant speed?

By producing what is known as an ON SPEED condition. This exists when the RPM is constant. Movement of the cockpit control has set the speeder spring at the desired RPM. The flyweights have positioned the pilot valve to direct oil to or from the propeller. This, in turn, has positioned the propeller blades at the pitch that absorbs the engine power at the RPM selected. When the moment the RPM balance occurs, the force of the flyweights equals the speeder spring load. This positions the pilot valve in the constant RPM position, with no oil flowing to or from the propeller.



OK, so we're flying along at constant RPM. What happens if the airplane begins to climb or engine power is decreased?

This results in an UNDERSPEED condition. Airspeed is reduced and, since the pitch of the propeller blades is too high, the engine starts to slow down. However, the instant this happens the flyweights will drop, causing the pilot valve to move down. Then oil flows from the propeller, reducing the pitch of the blades. This automatically increases the speed of the engine to maintain the former RPM setting.

ypical Filgnt Aititude

Piper Dakota (PA-28-236) Lycoming O-540-J3A5

Simplified Power Chart

- * With power settings less than 2300 RPM, maximum MP is limited
- * Above 8000' ISA, reduce climb speed 1 kt / 1000' and lean to 125F ROP

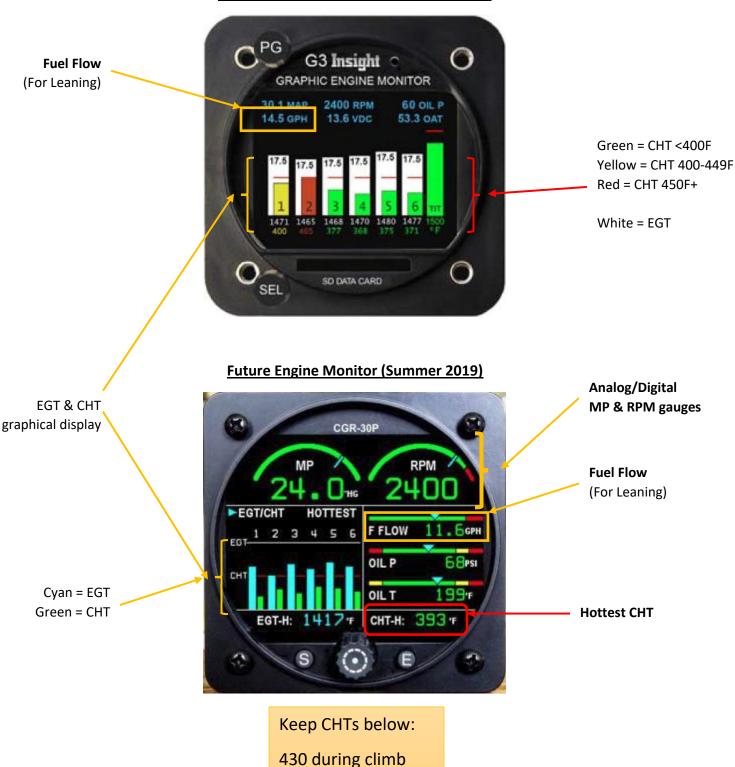
(Pressure) Altitude		65% Power @ <u>2300 RPM</u> 13.8 GPH			75% Power @ <u>2300 RPM</u> <i>16.5 GPH</i>		
2000		21.2" MP	120 KTAS		23.4" MP	130 KTAS	
3000		21.0" MP	122 KTAS		23.1" MP	132 KTAS	
4000		20.8" MP	124 KTAS		22.8" MP	135 KTAS	
5000		20.5" MP	126 KTAS		22.6" MP	137 KTAS	
6000		20.3" MP	128 KTAS		22.3" MP	138 KTAS	
7000		20.1" MP	130 KTAS		F.T.	140 KTAS	
8000		19.9" MP	132 KTAS				
9000		19.7" MP	134 KTAS		1		
10000		19.4" MP	136 KTAS				
		1					
21-20 ["] MP					23-22" MP		
subtract 0.2" / 1000'							

75% Power Compared to 65% Power

MP = 2" Higher
Fuel Burn = ~3 GPH Greater
TAS = 10 Knots Faster
Range = 70 NM less

Piper Dakota (PA-28-236) **Engine Monitor**

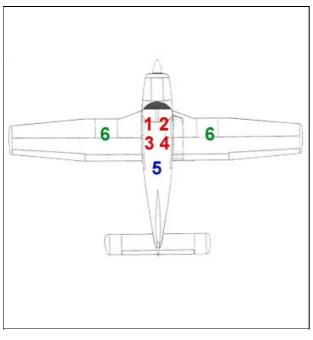
Current Engine Monitor (As of May 2019)



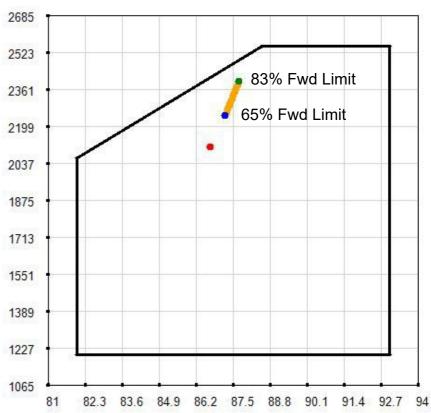
430 during climb

400 during cruise

Departure	
Arrival	
PIC	
SIC	
AC Performance	PA 28 181 ARCHER II
BEW/C.G.	1727.5/87
Pilot Seat	170
Seat 2	170
Seat 3	25
Seat 4	0
Baggage Compartment 1	15
Fuel Units	Gal.
Fuel Tank (Max. 48 gal.)	48
Fuel Burn	25
Max. Takeoff Weight	2550 lbs.
Under Gross Weight	154 lbs.
Zero Fuel Weight	2108 lbs.
Payload	380 lbs.
Time Stamp UTC -	05/04/2019 15:30



N455H



	Weight	Fwd Limit	C.G.	Aft Limit	% of MAC
Takeoff	2396	86.46	87.7	93	N/A
Landing	2246	84.47	87.21	93	N/A
Zero Fuel	2108	82.64	86.7	93	%

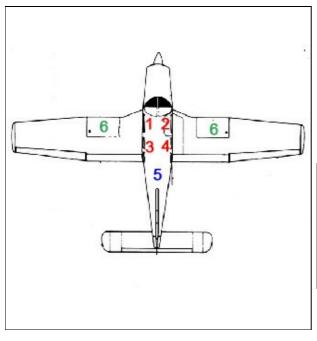
Chart Legend					
TakeOff C.G.					
Landing C.G.					
FuelBurn C.G.					
Zero Fuel Weight					
Envelope					

Takeoff = 48% Forward CG 52% Aft CG

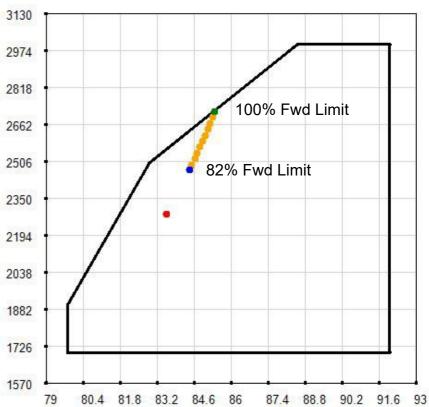
Landing = 53% Forward CG 47% Aft CG

ZFW = 57% Forward CG 43% Aft CG

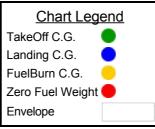
Departure	
Arrival	
PIC	
SIC	
AC Performance	PA 28 236 DAKOTA
BEW/C.G.	1904.7/83.2
Pilot Seat	170
Seat 2	170
Seat 3	25
Seat 4	0
Baggage Compartment 1	15
Fuel Units	Gal.
Fuel Tank (Max. 72 gal.)	72
Fuel Burn	41
Max. Takeoff Weight	3000 lbs.
Under Gross Weight	283 lbs.
Zero Fuel Weight	2285 lbs.
Payload	380 lbs.
Time Stamp UTC - 0	5/04/2019 15:27



N8107B



	Weight	Fwd Limit	C.G.	Aft Limit	% of MAC
Takeoff	2717	85.33	85.38	92	N/A
Landing	2471	82.75	84.42	92	N/A
Zero Fuel	2285	81.79	83.56	92	%

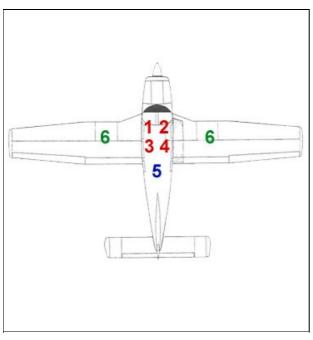


Takeoff = 54% Forward CG 46% Aft CG

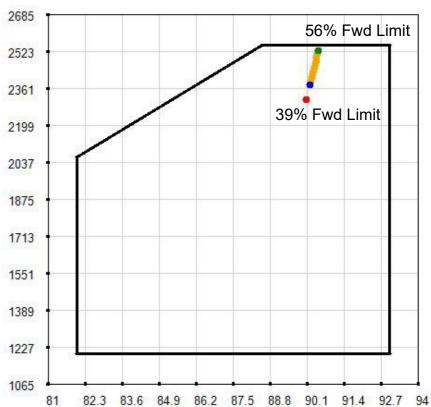
Landing = 62% Forward CG 38% Aft CG

ZFW = 69% Forward CG 31% Aft CG

Departure	
Arrival	
PIC	
SIC	
AC Performance	PA 28 181 ARCHER II
BEW/C.G.	1727.5/87
Pilot Seat	170
Seat 2	170
Seat 3	5
Seat 4	170
Baggage Compartment 1	70
Fuel Units	Gal.
Fuel Tank (Max. 48 gal.)	36
Fuel Burn	25
Max. Takeoff Weight	2550 lbs.
Under Gross Weight	21 lbs.
Zero Fuel Weight	2313 lbs.
Payload	586 lbs.
Time Stamp UTC -	05/04/2019 15:29



N455H



	Weight	Fwd Limit	C.G.	Aft Limit	% of MAC
Takeoff	2529	88.22	90.49	93	N/A
Landing	2379	86.23	90.21	93	N/A
Zero Fuel	2313	85.36	90.07	93	%

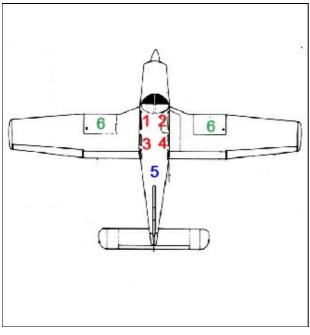


Takeoff = 23% Forward CG 77% Aft CG

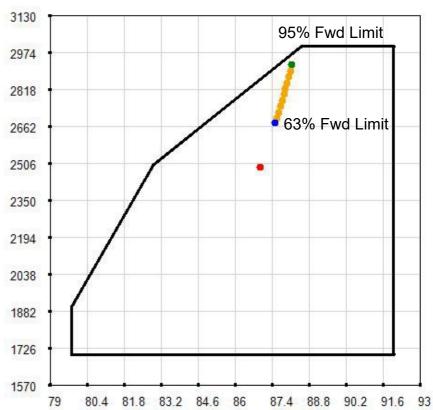
Landing = 25% Forward CG 75% Aft CG

ZFW = 27% Forward CG 73% Aft CG

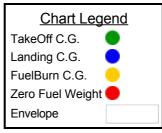
Departure	
Arrival	
PIC	
SIC	
AC Performance	PA 28 236 DAKOTA
BEW/C.G.	1904.7/83.2
Pilot Seat	170
Seat 2	170
Seat 3	5
Seat 4	170
Baggage Compartment 1	70
Fuel Units	Gal.
Fuel Tank (Max. 72 gal.)	72
Fuel Burn	41
Max. Takeoff Weight	3000 lbs.
Under Gross Weight	78 lbs.
Zero Fuel Weight	2490 lbs.
Payload	585 lbs.
Time Stamp UTC - 0	5/04/2019 15:26



N8107B



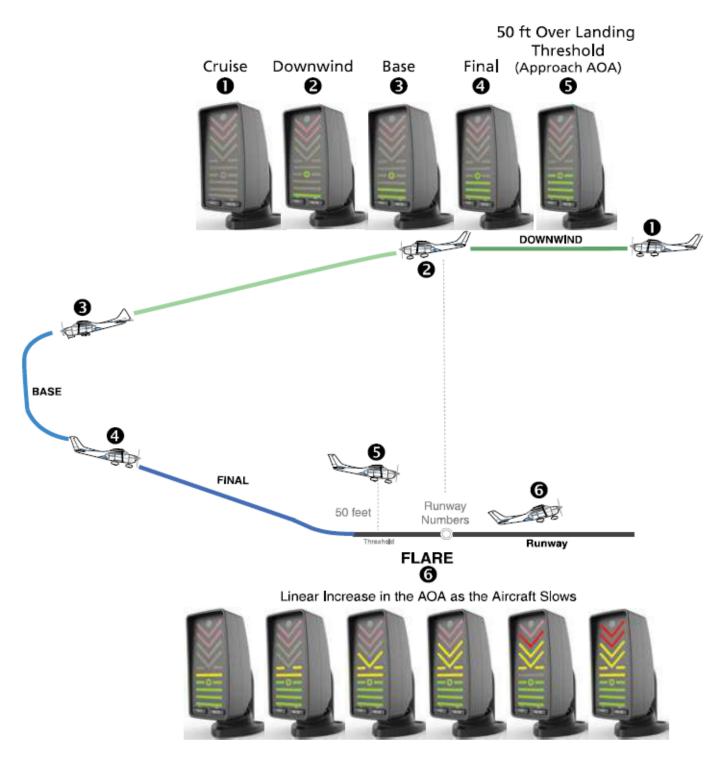
	Weight	Fwd Limit	C.G.	Aft Limit	% of MAC
Takeoff	2922	87.63	88.14	92	N/A
Landing	2676	84.87	87.51	92	N/A
Zero Fuel	2490	82.85	86.96	92	%



Takeoff = 32% Forward CG 68% Aft CG

Landing = 37% Forward CG 63% Aft CG

ZFW = 41% Forward CG 59% Aft CG



Example Indications for a Typical Decelerating Approach and Flare

S	Distance	667nm	204nm	87nm
Routes	Airports	KDXR - KLZU	KDXR - KFME	KDXR - KBID

	Weight	170	170	2	170	70	585
Loading	Location	Pilot	Front Pass.	Left Rear Pass.	Right Rear Pass.	Baggage	TOTAL

		Archer			
Payload	Weight Fu	Fuel/Weight	 	Airspeed	 0
Aircraft (55H)	1728		Altitude	65 %	75%
Loading	585		4000	116	124
Sub-Total	2313		0009	117	126
Remaining	237	39.5	7000		127
Fuel	36	216	8000	118	
Total	2529				
				Fndiirance	

GPH

510 85 72 432	2490	oading 585 4000	1905 Altitude		Dakota	75% 135 138 140	Airspee 65% 124 128 132	Altitude 4000 6000 7000 8000			
	510 85 72 43 <u>2</u>	510 85 72 432	585 4000 2490 6000 510 85 7000 72 432 8000	(B) 1905 Altitude 585 4000 2490 6000 72 432 8000	7B) Weight Fuel/Weight Altitude Altitude 2490 6000 72 432 8000					2922	otal

16.5

Endurance GPH 13.8 Time 5.2

Aircraft	Power	Trip Comparison Distance Time	parison Time	Fuel	Cost
	%59	204	5./ 1.6	73.4	\$266.85 \$266.85
2,0%		87	0.7	10.0	\$116.75
Dakora		299	5.3	81.5	\$800.54
	75%	204	1.5	25.1	\$250.17
		87	9.0	10.8	\$100.07
		299	6.7	62.2	\$676.59
	%59	204	1.8	19.3	\$213.66
v chor		87	0.7	9.7	\$83.09
ָ בובי בובי		299	6.4	68.5	\$640.98
	75%	204	1.7	21.1	\$201.79
		87	0.7	10.3	\$83.09
		299	-1.0	11.1	\$190.67
•	%59	204	-0.2	3.7	\$53.19
NO.		87	0.0	0.3	\$33.66
18410		299	-1.1	13.0	\$159.56
>	75%	204	-0.2	4.0	\$48.38
		87	-0.1	0.5	\$16.98